

Euler-Lagrange equation for  $r$  central force

$$\textcircled{1} \quad m \ddot{r}(t) = f[r(t)] + \frac{l^2}{m[r(t)]^3}$$

change variables  $u \equiv \frac{1}{r}$ ,  $du = -\frac{1}{r^2} dr$ ;  $\frac{dr}{du} = -r^2 = -\frac{1}{u^2}$

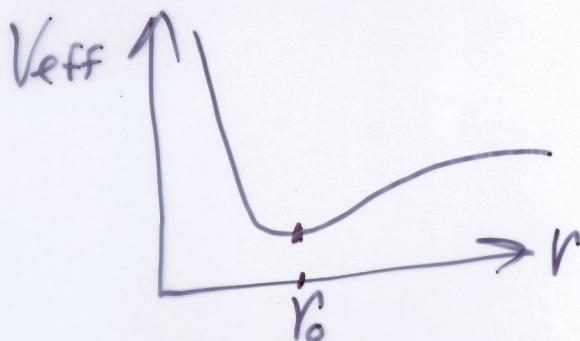
$$\textcircled{2} \quad \frac{d^2 u(\theta)}{d\theta^2} + u(\theta) = -\frac{m}{l^2} \frac{dV(\frac{1}{u})}{du}$$

Aside  $V(\frac{1}{u}) \neq V(u)$

e.g.  $V(r) = -\frac{k}{r} \neq V(u) = -\frac{k}{u} = -kr$

$\equiv$   
 $W(u)$

Examine small oscillations about equilibrium  
any function looks like a parabola near the minimum.



$\Rightarrow$  Look for Simple Harmonic Motion SHM

Define  $u(\theta) = u_0 + a \cos(\beta\theta)$

small  $a$ ,  $a \ll u_0$ ;  $u_0 \equiv \frac{1}{r_0}$

$$\frac{d^2 u}{d\theta^2} = -\beta^2 a \cos(\beta\theta)$$

$$\frac{dV}{du} = \frac{dV(r)}{dr} \frac{dr}{du} = (-f(r)) \left(-\frac{1}{u^2}\right) = \frac{f(r)}{u^2}$$

$$\textcircled{2} -\beta^2 a \cos(\beta\theta) + u_0 + a \cos(\beta\theta) = -\frac{m}{l^2} \frac{1}{u^2} \left[ f(r_0) + \left. \frac{df}{dr} \right|_{r_0} (r-r_0) + \dots \right]$$

At equilibrium  $r = r_0$ ,  $\dot{r} = 0$ ,  $\ddot{r} = 0$ .

$$0 = f(r_0) + \frac{l^2}{m r_0^3} \Rightarrow f(r_0) = -\frac{l^2 u_0^3}{m}$$

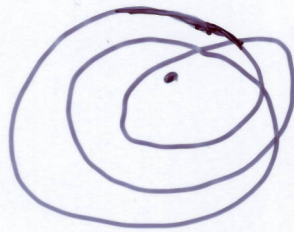
$$\begin{aligned} r - r_0 &= \frac{1}{u} - \frac{1}{u_0} = \frac{1}{u_0 + a \cos(\beta\theta)} - \frac{1}{u_0} = \frac{1}{u_0 \left[ 1 + \frac{a}{u_0} \cos(\beta\theta) \right]} - \frac{1}{u_0} \\ &= \frac{1}{u_0} \left[ 1 - \frac{a}{u_0} \cos(\beta\theta) \right] - \frac{1}{u_0} = -\frac{a}{u_0^2} \cos(\beta\theta) \end{aligned}$$

$$\begin{aligned} \frac{1}{u^2} &= \frac{1}{\left( u_0 + a \cos(\beta\theta) \right)^2} = \frac{1}{u_0^2 \left[ 1 + \frac{a}{u_0} \cos(\beta\theta) \right]^2} \\ &= \frac{1}{u_0^2} \left[ 1 - 2 \frac{a}{u_0} \cos(\beta\theta) \right] \end{aligned}$$

$$\begin{aligned} \textcircled{2} -\beta^2 a \cos(\beta\theta) + u_0 + a \cos(\beta\theta) &= -\frac{m}{l^2} \frac{1}{u_0^2} \left[ 1 - 2 \frac{a}{u_0} \cos(\beta\theta) \right] \cdot \\ &\quad \cdot \left[ -\frac{l^2 u_0^3}{m} + \left. \frac{df}{dr} \right|_{r_0} \left( -\frac{a}{u_0^2} \cos(\beta\theta) \right) \right] \\ &= u_0 - 2a \cos(\beta\theta) + \frac{m}{l^2 u_0^4} a \cos(\beta\theta) \left. \frac{df}{dr} \right|_{r_0} + O(a^2) \end{aligned}$$

$$\Rightarrow -\beta^2 + 3 = \frac{m}{l^2 u_0^4} \left. \frac{df}{dr} \right|_{r_0} = - \left( \frac{r}{f} \frac{df}{dr} \right) \Big|_{r_0}$$

As  $\theta$  goes from 0 to  $2\pi$ ,  $u$  goes through  $\beta$  cycles of radial oscillation. If  $\beta$  is rational  $\frac{p}{q}$  (reduced), the orbit will close after  $q$  revolutions,  $\theta = 2\pi q$

e.g. If  $\beta = \frac{5}{3} \Rightarrow$   closes after 3 orbits.

$$\beta^2 - 3 = \frac{d \ln f}{d \ln r} \Big|_{r_0}$$

$$d(\ln f) = \frac{df}{f}$$

$$d(\ln r) = \frac{dr}{r}$$

$$\Rightarrow d \ln f = (\beta^2 - 3) d \ln r \Rightarrow \ln f = (\beta^2 - 3) \ln r + A$$

$$\Rightarrow e^{\ln f} = f = e^{(\beta^2 - 3) \ln r + A} = r^{\beta^2 - 3} \cdot e^A$$

$$\Rightarrow f(r) = \frac{-k}{r^{3-\beta^2}} \quad \beta=1 \text{ gravity, } \beta=2 \text{ Hooke's Law Spring}$$

### Bertrand's Theorem

Any rational  $\beta = \frac{p}{q}$  will give closed orbits to first order in  $a$ . Only  $\beta = 1, 2$  to second order in  $a$ .

Mercury's perihelion precesses 5600 arcseconds per century. All but 43.1  $\frac{\text{arcsec}}{\text{cent}}$  are accounted for by other planets. General relativity gives the 43.1

$\frac{\text{arcsec}}{\text{cent}}$  Force  $f = \frac{-k}{r^2} \leftarrow 2,000,000,16$  for Mercury.



Keplerian Orbits  $f(r) = -\frac{k}{r^2}$   $V(r) = -\frac{k}{r}$

$$\theta = \theta_0 - \int \frac{du}{\sqrt{\frac{2mE}{l^2} + \frac{2mk}{l^2}u - u^2}}$$

$$\int \frac{dx}{\sqrt{\alpha + \beta x + \gamma x^2}} = -\frac{1}{\sqrt{-\gamma}} \arccos\left(\frac{-\beta - 2\gamma x}{\sqrt{\beta^2 - 4\alpha\gamma}}\right)$$

$$\alpha = \frac{2mE}{l^2} \quad \beta = \frac{2mk}{l^2} \quad \gamma = -1$$

$$\theta = \theta_0 - \arccos\left[\frac{\frac{l^2 u}{mk} - 1}{\sqrt{1 + \frac{2El^2}{mk^2}}}\right] \quad \text{solve for } u$$

$$u = \frac{1}{r} = \frac{mk}{l^2} \left[ 1 + \sqrt{1 + \frac{2El^2}{mk^2}} \cos(\theta - \theta_0) \right]$$

$r_{\min}$  at  $\theta = \theta_0$   
(perihelion)

$r_{\max}$  at  $\theta = \theta_0 + \pi$   
(aphelion)

↑ turning points or apsides ↑

conic section with  
one focus at the  
origin

$$\frac{1}{r} = C \left[ 1 + \underset{\substack{\uparrow \\ \text{eccentricity}}}{e} \cos(\theta - \theta_0) \right]$$

$$e = \sqrt{1 + \frac{2El^2}{mk^2}}$$

$e > 1$	$E > 0$	—	hyperbola
$e = 1$	$E = 0$	—	parabola
$e < 1$	$E < 0$	—	ellipse
$e = 0$	$E = -\frac{mk^2}{2l^2}$	—	circle

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Virial Theorem  $n = -2$  :  $\langle T \rangle = -\frac{1}{2} \langle V \rangle$

In a circular orbit,  $T$  and  $V$  are constants

$$T = -\frac{1}{2}V$$

$$E = T + V = -\frac{1}{2}V + V = \frac{1}{2}V = \frac{-k}{2r_0} = -\frac{mk^2}{2l^2}$$

$$f(r_0) = -\frac{l^2}{m r_0^3} = -\frac{k}{r_0^2}$$


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### Elliptical Orbits

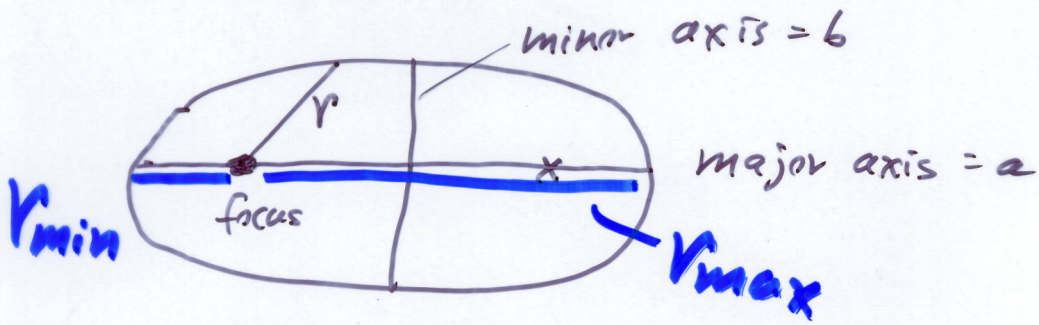
At apsides,  $r_{\min}$  and  $r_{\max}$ ,  $\dot{r} = 0$  instantaneously

$$E = T + V = \frac{1}{2}m\dot{V}_r^2 + \frac{1}{2}m\dot{V}_\theta^2 + V$$

$$= \underbrace{\frac{1}{2}m\dot{r}^2}_0 + \frac{l^2}{2mr^2} - \frac{k}{r}$$

$$\Rightarrow \frac{l^2}{2mr^2} - \frac{k}{r} - E = 0 \quad \text{quadratic equation in } r$$

$$r_{\max/\min} = \frac{-\frac{k}{E} \pm \sqrt{\frac{k^2}{E^2} + \frac{2l^2}{mE}}}{2}$$



$$\text{Semi-major axis} = \frac{a}{2} = \frac{r_{\min} + r_{\max}}{2} = \frac{-k}{2E}$$

$$\text{eccentricity } e = \sqrt{1 - \frac{l^2}{mka}}$$

$$\Rightarrow \frac{l^2}{mk} = a(1 - e^2)$$

$$r = \frac{a(1 - e^2)}{1 + e \cos(\theta - \theta_0)} \quad \leftarrow \text{formula for ellipse}$$

Kepler's first law