

1) Sun & Earth

Part a)

```
In[7]:= Clear["Global`*"]
```

```
In[8]:= (* The answers *)
```

```
mSunAnswer = 2 × 1030;
```

```
mEarthAnswer = 6 × 1024;
```

```
In[1]:= earthOrbit = 1.49 × 108 km;  
moonOrbit = 3.8 × 105 km; m
```

```
Out[2]= m
```

```
In[12]:= eq1 = {  $\frac{G m_1 m_2}{r^2} == m_1 \frac{v^2}{r}$ ,  $v == \frac{2 \pi r}{\text{period}}$  };
```

```
In[13]:= sol = Solve[eq1, {m2, v}][[1]]
```

```
Out[13]= {m2 →  $\frac{4 \pi^2 r^3}{G \text{period}^2}$ , v →  $\frac{2 \pi r}{\text{period}}$ }
```

```
In[14]:= mass = m2 /. sol
```

```
Out[14]=  $\frac{4 \pi^2 r^3}{G \text{period}^2}$ 
```

```
In[15]:= mSun = mass /. {r → earthOrbit, period → 365.25 days}
```

```
Out[15]=  $\frac{9.78899 \times 10^{20} \text{ km}^3}{\text{days}^2 G}$ 
```

```
In[16]:= mEarth = mass /. {r → moonOrbit, period → 27.3 days}
```

```
Out[16]=  $\frac{2.9066 \times 10^{15} \text{ km}^3}{\text{days}^2 G}$ 
```

```
In[17]:=  $\frac{m_{\text{Sun}}}{m_{\text{Earth}}}$  // ScientificForm
```

```
Out[17]/ScientificForm= 3.36785 × 105
```

```
In[18]:=  $\frac{m_{\text{SunAnswer}}}{m_{\text{EarthAnswer}}}$  // N // ScientificForm
```

```
Out[18]/ScientificForm= 3.33333 × 105
```

Part b) Measure relative to the center of the sun

```
In[28]:= Clear["Global`*"]

In[37]:= mSun = 2.0 × 1030;
mEarth = 6.0 × 1024;
dEarthSun = 149. × 109;
radiusSun = 696 × 106;

In[42]:= reducedMass =  $\frac{m_{\text{Sun}} m_{\text{Earth}}}{m_{\text{Sun}} + m_{\text{Earth}}}$ 

Out[42]:= 5.99998 × 1024

In[43]:=  $\frac{\text{reducedMass}}{m_{\text{Sun}}}$ 

Out[43]:= 2.99999 × 10-6
```

Part c) Measure relative to the center of the sun

```
In[28]:= Clear["Global`*"]

In[37]:= mSun = 2.0 × 1030;
mEarth = 6.0 × 1024;
dEarthSun = 149. × 109;
radiusSun = 696 × 106;

In[36]:= rcms =  $\frac{m_{\text{Sun}} \times 0 + m_{\text{Earth}} d_{\text{EarthSun}}}{m_{\text{Sun}} + m_{\text{Earth}}}$ 

Out[36]:= 446 999.

In[41]:=  $\frac{\text{rcms}}{\text{radiusSun}}$ 

Out[41]:= 0.000642239
```

2) Orbits

```
In[55]:= Clear["Global`*"]

In[56]:= ? PolarPlot
```

Out[56]=

Symbol



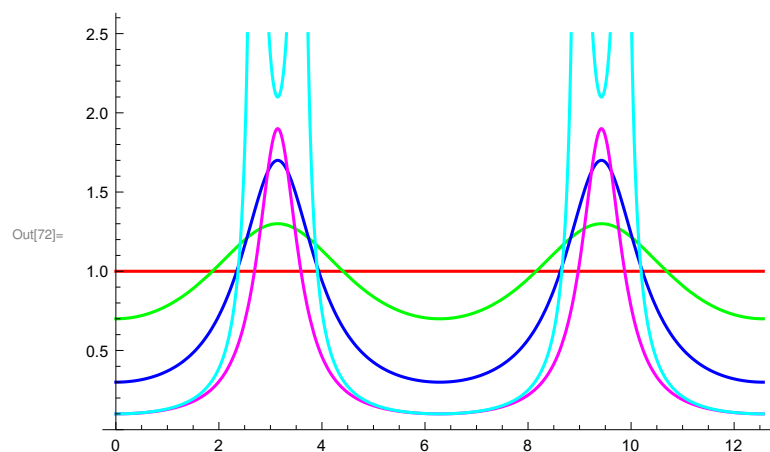
PolarPlot [$r, \{\theta, \theta_{\min}, \theta_{\max}\}$] generates a polar plot of a curve with radius r as a function of angle θ .
 PolarPlot [$\{r_1, r_2, \dots\}, \{\theta, \theta_{\min}, \theta_{\max}\}$] makes a polar plot of curves with radius functions $r_1, r_2, \dots$



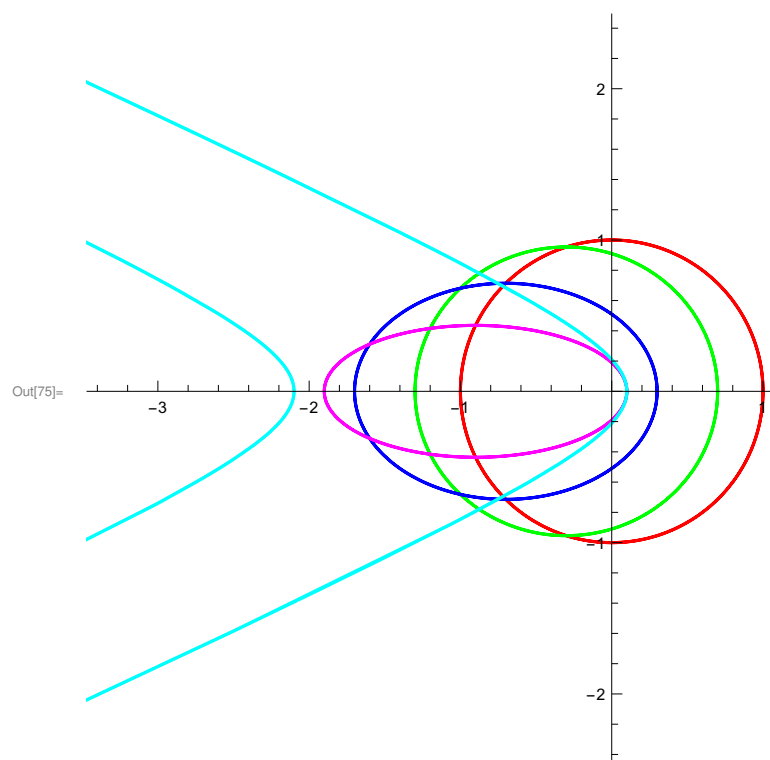
In[57]:= $r[\theta_, e_ : 0, a_ : 1] = \frac{a(1 - e^2)}{1 + e \cos[\theta]}$

Out[57]= $\frac{a(1 - e^2)}{1 + e \cos[\theta]}$

In[72]:= `Plot[{r[θ, 0.0], r[θ, 0.3], r[θ, 0.7], r[θ, 0.9], r[θ, 1.1]} // Abs // Evaluate, {θ, 0, 4 π}, PlotStyle → {Red, Green, Blue, Magenta, Cyan}, PlotRange → {0, Automatic}]`



In[75]:= `PolarPlot[{r[θ, 0.0], r[θ, 0.3], r[θ, 0.7], r[θ, 0.9], r[θ, 1.1]} // Abs // Evaluate, {θ, 0, 4 π}, PlotStyle → {Red, Green, Blue, Magenta, Cyan}, PlotRange → Automatic]`



```
In[77]:= eq1 = D[r[θ, e, a], θ] == 0
```

$$\text{Out[77]} = \frac{a e (1 - e^2) \sin[\theta]}{(1 + e \cos[\theta])^2} == 0$$

```
In[79]:= Solve[eq1, θ] /. {C[_] -> 0}
```

```
Out[79]= {{θ -> 0}, {θ -> π}}
```

3) Small Meteoroid

```
In[3]:= Clear["Global`*"]
```

Part a)

$$\text{In[10]} := \text{eq1} = \frac{G M m}{r^2} == m \frac{v^2}{r};$$

```
vSol = Solve[eq1, v][[2]]
```

$$\text{Out[11]} = \left\{ v \rightarrow \frac{\sqrt{G} \sqrt{M}}{\sqrt{r}} \right\}$$

```
In[12]:= angMom = m r v /. vSol
```

$$\text{Out[12]} = \sqrt{G} m \sqrt{M} \sqrt{r}$$

$$\text{In[14]} := \text{energy0} = \frac{1}{2} m v^2 - \frac{G M m}{r} /. vSol$$

$$\text{Out[14]} = -\frac{G m M}{2 r}$$

Part b) After exploding, v is the same, the mass of the Sun “M” reduces from M -> κ M, where we will determine κ. If it is just barely bound, the new energy will be zero.

$$\text{In[15]} := \text{energy1} = \frac{1}{2} m v^2 - \frac{G \kappa M m}{r} /. vSol$$

$$\text{Out[15]} = \frac{G m M}{2 r} - \frac{G m M \kappa}{r}$$

```
In[18]:= Solve[energy1 == 0, κ][[1]]
```

$$\text{Out[18]} = \left\{ \kappa \rightarrow \frac{1}{2} \right\}$$

Part 3) We will conserve energy and angular Momentum

In[45]:= **energy2 = energy1 /. {κ → 3/4}**

$$\text{Out[45]} = -\frac{G m M}{4 r}$$

In[38]:= **angMom == m rMax vMax**

$$\text{Out[38]} = \sqrt{G} m \sqrt{M} \sqrt{r} == m r_{\text{Max}} v_{\text{Max}}$$

In[46]:= **kineticEnergyMax = $\frac{1}{2} m v_{\text{Max}}^2$**

$$\text{Out[46]} = \frac{m v_{\text{Max}}^2}{2}$$

In[47]:= **potentialEnergyMax = - $\frac{G \kappa M m}{r_{\text{Max}}}$ /. {κ → 3/4}**

$$\text{Out[47]} = -\frac{3 G m M}{4 r_{\text{Max}}}$$

In[49]:= **totalEnergy = kineticEnergyMax + potentialEnergyMax**

$$\text{Out[49]} = -\frac{3 G m M}{4 r_{\text{Max}}} + \frac{m v_{\text{Max}}^2}{2}$$

In[50]:= **eqs = {angMom == m rMax vMax, energy2 == totalEnergy}**

$$\text{Out[50]} = \left\{ \sqrt{G} m \sqrt{M} \sqrt{r} == m r_{\text{Max}} v_{\text{Max}}, -\frac{G m M}{4 r} == -\frac{3 G m M}{4 r_{\text{Max}}} + \frac{m v_{\text{Max}}^2}{2} \right\}$$

In[52]:= **Solve[eqs, {rMax, vMax}][[1]]**

$$\text{Out[52]} = \left\{ r_{\text{Max}} \rightarrow 2 r, v_{\text{Max}} \rightarrow \frac{\sqrt{G} \sqrt{M}}{2 \sqrt{r}} \right\}$$

4) Haleys Comet

In[200]:= **Clear["Global`*"]**

In[201]:= **kepler = period² == α r³**

$$\text{Out[201]} = \text{period}^2 == r^3 \alpha$$

In[202]:= **eqEarth = kepler /. {period → 1 year, r → 1 au}**

$$\text{Out[202]} = \text{year}^2 == \text{au}^3 \alpha$$

In[203]:= **eqComet** = **kepler** /. {period → 75.6 year, r → $\frac{(0.570 + rMax)}{2}$ au}

Out[203]= $5715.36 \text{ year}^2 == \frac{1}{8} \text{ au}^3 (0.57 + rMax)^3 \alpha$

In[204]:= **eqs** = {**eqEarth**, **eqComet**}

Out[204]= $\left\{ \text{year}^2 == \text{au}^3 \alpha, 5715.36 \text{ year}^2 == \frac{1}{8} \text{ au}^3 (0.57 + rMax)^3 \alpha \right\}$

In[205]:= **Solve**[eqs, { α , rMax}]

Out[205]= $\left\{ \left\{ \alpha \rightarrow \frac{\text{year}^2}{\text{au}^3}, rMax \rightarrow -18.4492 - 30.9677 i \right\}, \right.$
 $\left. \left\{ \alpha \rightarrow \frac{\text{year}^2}{\text{au}^3}, rMax \rightarrow -18.4492 + 30.9677 i \right\}, \left\{ \alpha \rightarrow \frac{\text{year}^2}{\text{au}^3}, rMax \rightarrow 35.1884 \right\} \right\}$

5) Escape Velocity

In[180]:= **Clear**["Global`*"]

In[181]:= **mSun** = 2×10^{30} ;
mEarth = 6×10^{24} ;
rEarth = 6000×10^3 ;

In[184]:= **eq1** = $\frac{-G M m}{r} + \frac{1}{2} m v^2$

Out[184]= $-\frac{G m M}{r} + \frac{m v^2}{2}$

In[185]:= **sol** = **Solve**[eq1 == 0, v][[2]]

Out[185]= $\left\{ v \rightarrow \frac{\sqrt{2} \sqrt{G} \sqrt{M}}{\sqrt{r}} \right\}$

In[196]:= **values** = {**G** → 6.67×10^{-11} , **c** → 3.0×10^8 }

Out[196]= {**G** → 6.67×10^{-11} , **c** → $3. \times 10^8$ }

In[187]:= **v** /. **sol** /. **values** /. {**r** → **rEarth**, **M** → **mEarth**}

Out[187]= 11549.9

In[189]:= **eq2** = **c** == **v** /. **sol**

Out[189]= **c** == $\frac{\sqrt{2} \sqrt{G} \sqrt{M}}{\sqrt{r}}$

In[192]:= **cSol = Solve[eq2, r][[1]]**

Out[192]= $\left\{ r \rightarrow \frac{2 G M}{c^2} \right\}$

In[198]:= **r /. cSol /. values /. {M -> mEarth}**

Out[198]= **0.00889333**

In[199]:= **r /. cSol /. values /. {M -> mSun}**

Out[199]= **2964.44**